















































































## FMCAD'2012, Cambridge Formal methods in Avionics tutorial

22<sup>th</sup> october 2012



Marc Pantel – ACADIE team Assistance à la Certification d'Applications DIstribuées et Embarquées

	Safe MDE concerns
DO178/ED12 safety standards	<ul> <li>Main purpose : Safety critical systems</li> <li>Main approach : formal specification and verification</li> <li>Problems : expressiveness, decidability, completeness, consistency</li> </ul>
Application to Code generation tools Application to Static analysis tools Synthesis : Open questions ?	<ul> <li>Proposals : Raise abstraction</li> <li>Higher level programming languages and frameworks</li> <li>Domain specific (modeling) languages</li> <li>easy to access for end users</li> <li>with a simple formal embedding</li> <li>with automatic verification tools</li> <li>with usefull validation and verification results</li> <li>that are accepted by certification authorities</li> </ul> Needs : <ul> <li>methods and tools to ease their development</li> <li>algebraic and logic theoretical fondations</li> <li>proof of transformation and verification correctness</li> <li>links with certification/qualification</li> </ul>
ACADI	E team (2/38)

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ACAD	DIE team (2/38)









**ACADIE team** 

CORRECTIONS OF CONTRACTORS OF CONTRA	<ul> <li>DO-178/ED-12 safety standards : Certification</li> <li>Key issue : Choose the strategy and technologies that will minimize risks         Assessment : Stochastic for system, Zero-default for Software         Arocess and test-centered approach         Definition of a precise process (development/verification)         AC-DC test coverage for DAL A truth-table lines of sub-expressions in conditions (some can be merged)         Asymmetry with independence argument : several activities (and products) by different teams, with different tools,</li> </ul>	
ACADIE	team (6/3	8)






















**ACADIE team** 

	Formal methods
ACADIE DO178/ED12 safety standards Application to Code	<ul> <li>A formal method must be correctly defined, justified and appropriate</li> <li>Correctly defined : precise, unambiguous, mathematically defined syntax and semantics</li> <li>Justified : Sound (never assert a false property)</li> <li>Appropriate : All assumptions required for the formal analysis should be described and justified</li> </ul>
generation tools Application to Static analysis tools Synthesis : Open	<ul> <li>Requirement formalization correctness</li> <li>Formal analysis can replace :         <ul> <li>Review and analysis objectives</li> <li>Conformance tests versus HLR and LLR</li> <li>Robustness tests</li> <li>Compatibility with the hardware (WCET,)</li> </ul> </li> </ul>
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ACAD	IE team (14 / 38 )

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ACADIE team	( 14 / 38 )

## DO-178B/ED-12B standards : Qualification 2iT ACADIE Apply to the tools the same rules as the developped system at the same level DO178/ED12 safety Not really adequate : Additional documents (CAST) were provided standards Tool Operational Requirements (TOR) : **Application to** Tool user point of view (similar to System requirements – HLR) Code generation Tool Requirements (TR) : tools Tool implementor point of view (LLR) Application to Tool kind : analysis tools • Development tools : Synthesis : Open Tools whose output is part of airborne software and thus can introduce questions? errors (same constraints as the developed system). Verification tools : Tools that cannot introduce errors, but may fail to detect them (much softer constraints : black box V & V). No proof of error absence category **ACADIE** team

(15/38)

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**ACADIE team** 

















(18/38)











DO178/ED12 safety standards

Application to Code generation tools

Application to Static analysis tools

Synthesis : Open questions ?

## What are :

**Open questions ?** 

- User requirement (TOR) for a transformation/verification?
- Developer requirement (TR) for a transformation/verification?
- Formal specification for a transformation/verification?
- Test coverage for a transformation/verification?
- Test oracle for a transformation/verification ?
- Qualification constraint for transformation/verification languages?
- Best strategy for tool verification (once vs at each use)?









CNRS - INPT - UPS - UT	Early feedbacks
Image: Constraint of the second se	<ul> <li>Separation of concerns : <ul> <li>Industrial partners : Specification, Implementation, Implementation verification (mainly syntactic)</li> <li>Academic partners : Specification verification (semantics)</li> </ul> </li> <li>Very good subcontracting capabilities</li> <li>Almost no technology constraints on the industrial partner (classical technologies)</li> <li>Good scalability</li> <li>Rely on already mandatory traceability links (formalized and verified)</li> <li>Easy to analyse syntactic error reports</li> <li>Enables to modify generated code and links</li> <li>Parallel work between syntactic and semantics concerns</li> </ul>
ACAD	IE team (27 / 38



CNRS - INPT - UPS - U	Static analysis tools
ACADIE	<ul> <li>Several kind of tools</li> </ul>
DO178/ED12 safety standards	<ul> <li>Qualitative and quantitative properties</li> <li>Fixed or user defined properties</li> <li>Semantic abstraction or Proof technologies</li> </ul>
Application to Code generation tools Application to Static	<ul> <li>Common aspects : Common pre-qualification</li> <li>Product (source of binary code) reader : fully common ?</li> <li>Configuration (properties,) reader : partly common</li> <li>Result writer and browser : partly common ?</li> </ul>
analysis tools Synthesis : Open questions ?	<ul> <li>Split the verification tool in a sequence of elementary activities</li> <li>Common ones (pre-qualification could be shared)</li> <li>Technology specific ones</li> <li>Easier to specify, to validate and to verify</li> <li>Can be physical or virtual (produce intermediate results even in a single tool)</li> </ul>

ACADIE team

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	Deductive kind	
Image: Constraint of the second sec	<ul> <li>Produce proof obligations (weakest precondition, verification condition,)</li> <li>Check the satisfaction of proof obligations <ul> <li>Proof term rewriting to simpler language</li> <li>Split to different sub-languages (pure logic, arithmetic,)</li> <li>Apply heuristics to produce a proof term</li> <li>Check the correctness of the proof term</li> <li>Produce failure feedback or proof certificate (related to product and its standard semantics)</li> </ul> </li> <li>Produce user friendly feedback</li> </ul>	
ACAD	IE team ( 32 / 38	)
























National Aeronautics and Space Administration

#### **Aviation Safety Program**

System-Wide Safety and Assurance Technologies (SSAT) Project Assurance of Flight Critical Systems

A window into AFCS

Dr. Guillaume Brat, NASA Ames Research Center

October 22, 2012

www.nasa.gov

## Impact: Cost, and Constraints on Innovation



Lines of Code

1.2M	
1.7M	
3.6M	
4M	
6.5M	
5.7M	
100M	
	1.2M 1.7M 3.6M 4M 6.5M 5.7M 100M



NASA Study Flight Software Complexity, 4/23/2009

10/22/12

System

Boehm, B. 1981 Software Engineering Economics, as cited in DAA, 2008

## **Certification Aspect**





# **IKOS**

#### **Inference Kernel of Open Static analyzers**

Accomplishment: we have designed and implemented IKOS, a static analysis framework, which allows the custom design of mathematically sound analyzers, e.g., no false negatives, producing less than 10% false positives.

We implemented a generalization of the array-out-of-bound  $\checkmark$ accesses analysis

✓ New Gauge abstraction domain (CAV 2012)

✓ The framework is being processed to be released under a NOSA license

 $\checkmark$  Initial experiments show that we have achieved less than 10% false positive on embedded system code.

No 3 on the CWE/ SANS Top 25 Most **Dangerous Software** Errors (MITRE)

code	Size	Analysis time	Precision
Paparazzi	35 KLOC	22s	99%
Gen2	22 KLOC	1mn03s	98%
FLTz	144 KLOC	10mn30s	91%
Arduplane	278 KLOC	6mn30s	94%



## **Recent Advances**



- The analysis of the OpenSSH code required a sophisticated abstraction based on higher-dimensional convex polyhedra:
  - Combinatorial explosion
  - Brittle abstraction
  - A nightmare to analyze 700 lines of code
- Developed a new abstraction in IKOS: the Gauge Domain
  - Published in CAV 2012
  - Accuracy comparable to convex polyhedra
  - Analyzes 150 KLOC in minutes
  - Scales linearly with the size of the code
  - Commercial analyzer (PolySpace) takes hours on the same code

## Gauges



• In our experience with analyzing large NASA codes, we have observed that most of the time, the value of a scalar variable inside a loop nest was entirely determined by the control structure in terms of symbolic bounds of the form  $a_0 + a_1\lambda_1 + \dots + a_k\lambda_k$ , where  $\lambda_1, \dots, \lambda_k$  denote loop counters and  $a_1, \dots, a_k$  are integer coefficients.

# **Analysis of Floating-Point Computations**



- Challenging problem
- Solutions exist for a very small class of codes
  - ASTREE analyzer developed in France for Airbus
  - In practice only works for Airbus code
- Ongoing development of abstractions for floating-point computations in IKOS
  - Broader class of codes (UAVs, ground control)
  - Performance is the key issue



# **Analysis of Autocoded Systems**

- Model-based development:
  - Specify a controller in a mathematical modeling environment (MATLAB/Simulink)
  - Do all the verification at the model level
  - Automatically generate code from the model
- Question:
  - Does the generated code verify the same properties established at the model level?
  - Use static analysis to answer this question automatically

# **Example: Stability of Control Systems**

- Lyapunov theory
- Well understood at the model level: find an ellipsoidal invariant
- Does this hold for the generated code?
  - Discretization
  - Floating-point arithmetic
- Development of a suitable domain of ellipsoids for static analysis





# **IKOS: Applications**

- Applied to UAV autopilots (40 to 250 KLOC)
  - Few inconclusive reports (< 2%)</li>
- Running on LADEE (Lunar Atmosphere and Dust Environment Explorer) flight software
- Ongoing:
  - Scientific computation code (Corey Ippolito)
  - Image processing code for GOES-R (Geostationary Operational Environmental Satellite R-Series)

## compositional verification





- performed at design time
  - local properties guarantee P
- decomposition can be performed manually (e.g., for system architectures)
- we provide automated techniques based on learning component interfaces

•

local P<sub>4</sub>

individual components can be checked against local properties using model checking or testing

10/22/12

local P<sub>3</sub>

## what local properties express (CEV)



- constraints on inputs for correct operation
- constraints on sequences of method / service invocations

#### verification of separation assurance systems



• it is high in complexity (non-linear math and heuristics) with over 50k lines of Java; it has a complex interaction mechanism via callbacks into the Airspace Concept Evaluation System or ACES, a simulator that captures the key feedback response mechanism of the National Airspace System (NAS)









#### our efforts



- abstract ACES –trajectory computation
- generate conflicts
- work with Code A to identify properties:
  - "If the resolution produced for a conflict results in a secondary conflict, then the time to loss of separation for the secondary conflict should be greater than the time to loss of separation for the original conflict"
- developing advanced automated testing techniques that ensure coverage of all the paths of the AutoResolver
- working on developing and connecting all the parts of the compositional verification picture 10/22/12

## Conclusions



- Stable research program under ARMD for developing safety assurance techniques for flight-critical systems
- The main focus is on formal methods
  - Abstract interpretation
  - Compositional verification
  - Advanced test generation techniques
- Collaboration with
  - NASA Langley
  - Industry
  - International partners (ONERA, IRIT, may be VERIMAG)

# Formal Methods for Aerospace Applications: A control engineer's perspective

Eric Feron Dutton/Ducoffe Professor of Aerospace Engineering Georgia Institute of Technology

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# Take-Home Message

Safety-critical embedded software design best tackled through proper specification, followed by automatic coding of specs AND their semantics



# Analyze and Design Early

- Most errors arise during specification of software, not coding.
- Allow the engineer to specify, analyze, then auto-code.
- SCADE/Esterel Technologies, Picture2code/Pratt & Whitney, Realtime Workshop/Mathworks, Geneauto/ENSEEIHT, Gryphon/Rockwell-Collins.



# A simple control example



$$\frac{d}{dt} \begin{bmatrix} x \\ \dot{x} \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -1 & 0 \end{bmatrix} \begin{bmatrix} x \\ \dot{x} \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u, \quad x(0) = x_0, \dot{x}(0) = \dot{x}_0$$
$$y = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} x \\ \dot{x} \end{bmatrix}.$$







$$\begin{split} \tilde{y}(t) &= \mathbf{SAT}(y(t)), \\ u(s) &= 128 \frac{s+1}{s+0.1} \frac{s/5+1}{s/50+1} \tilde{y}(s), \end{split}$$





# **Controller implementation**



## Code-level analyses of control software

- Most significant contribution is from Patrick Cousot's research group at Ecole Normale Superieure, Paris.
- Abstract interpretation aims at capturing semantics of programs
- Most important application is ASTREE analyzer for Airbus A380 control code.
- From Feret, "Static Analysis of Digital Filters", 2004 (also with ASTREE).

Static Analysis of Digital Filters 43

A simplified second order filter relates an input stream  $E_n$  to an output stream defined by:

 $S_{n+2} = aS_{n+1} + bS_n + E_{n+2}$ . Thus we experimentally observe, in Fig. 4, that starting with  $S_0 = S_1 = 0$  and provided that the input stream is bounded, the pair  $(S_{n+2}, S_{n+1})$  lies in an ellipsoid. Moreover, this ellipsoid is attractive, which means that an orbit starting out of this ellipsoid, will get closer of it. This behavior is explained by Thm. 5.



Fig. 4. Orbit.



# A Paradigm Shift Enabled by Good **Specification Analyses**

(auto) Code analyzer



Credible autocoder (a la Rinard)



School of Aerospace Engineering

# Desirable attributes of "system proofs"

- Must be expressive enough to tell nontrivial statements about system
- Must speak the language of system representation, eg: "IEEE Transactions on Automatic Control proofs" written in natural language (one wonders...), "Simulink proofs" expressed in Simulink, "Program proofs" expressed in formal languages.
- Must be "elementary enough" to be easily checked wherever necessary.






### Back to the Example

The control-systemic way:

$$\begin{aligned} x_{c,k+1} &= \begin{bmatrix} 0.499 & -0.050 \\ 0.010 & 1.000 \end{bmatrix} x_{c,k} + \begin{bmatrix} 1 \\ 0 \end{bmatrix} \mathbf{SAT}(y_k) \\ u_k &= -\begin{bmatrix} 564.48 & 0 \end{bmatrix} x_{c,k} + 1280 \mathbf{SAT}(y_k) \end{aligned}$$

Assume the controller state is initialized at  $x_{c,0} = 0$ 

What range of values could be reached by the state  $x_{c,k}$  and the control variable  $u_k$ ?

There is a variety of options, including computation of -1 norms.

A Lyapunov-like proof (from Boyd *et al.*, Poola):

The ellipsoid 
$$\mathcal{E}_P = \{ x \in \mathbf{R}^2 \mid x^T P x \le 1 \}.$$
  $P = 10^{-3} \begin{bmatrix} 0.6742 & 0.0428 \\ 0.0428 & 2.4651 \end{bmatrix}.$ 

is invariant. None of the entries of *x* exceeds 7 in size.



### A proof for control people

 $\forall t, \ x^T P x \leq 1 \text{ is equivalent to } x_k^T P x_k \leq 1 \Rightarrow x_{k+1}^T P x_{k+1} \leq 1$ 

Or  $(Ax + Bw)^T P(Ax + Bw) \leq 1$  whenever  $x^T Px \leq 1$  and  $w^2 \leq 1$ 

True if there exists  $\mu$  such that  $(Ax+Bw))^T P(Ax+Bw) - \mu x^T Px - (1-\mu)w^2 < 0$ , (\*) a tautology.

Indeed a linear combination of (\*) and  $x^T P x \leq 1$  and  $w^2 \leq 1$  yields the desired property.

 $P \text{ that works is } P = 10^{-3} \begin{bmatrix} 0.6742 & 0.0428 \\ 0.0428 & 2.4651 \end{bmatrix}, \text{ with } \mu = 0.9991 \text{ and tautology}$  $(*) \text{ is } 10^{-3} \begin{bmatrix} x \\ w \end{bmatrix}^T \begin{bmatrix} -0.5044362 & -0.0135878 & 0.3374606 \\ -0.0135878 & -0.0003759 & 0.00909 \\ 0.3374606 & 0.00909 & -0.2258 \end{bmatrix} \begin{bmatrix} x \\ w \end{bmatrix} \le 0.$ 



### Simulink, Discrete Time Formal Semantics



```
{true}
1: A = [0.4990, -0.0500; 0.0100, 1.0000];
{true}
2: C = [-564.48, 0];
                                                        Commented code
{true}
3: B = [1;0]; D=1280
{true}
4: x = zeros(2,1);
\{x \in \mathcal{E}_P\}
5: while 1
\{x \in \mathcal{E}_P\}
6: y = fscanf(stdin,"%f")
\{x \in \mathcal{E}_P\}
7: y = max(min(y,1),-1);
\left\{x\in \mathcal{E}_P, \ y^2\leq 1\right\}
8: u = C*x+D*y;
\{x \in \mathcal{E}_P, u^2 \le 2(CP^{-1}C^T + D^2), y^2 \le 1\}
9: fprintf(stdout,"%f\n",u)
\{x \in \tilde{\mathcal{E}}_P, y^2 \leq 1, (Ax + By)^T P(Ax + By) - 0.01x^T Px - 0.99y^2 \leq 0\}
skip
\left\{Ax + By \in \mathcal{E}_P, \ y^2 \le 1\right\}
10: x = A*x + B*y;
\{x \in \mathcal{E}_P\}
11: end
```



## Front End: Formal comment writing



- ANSI/ISO C Specification Language (ACSL) can be used to formally comment C programs and can be handled by Frama-C.
- Start from Simulink
- End with commented C code



### A prototype front-end built on Gene-Auto

- Thank you Marc Pantel, Arnaud Dieumegard, Andres Toom



### Back End: Verification of Code Semantics





# A physical example: 3 DOF helicopter



# And it still works!!!





Daniel Guggenheim School of Aerospace Engineering



## F-18 replica from Rockwell-Collins



http://www.youtube.com/watch?v=QJkIONTzbNM



#### Application to Industrial Example: F/A-18 UAV

Complexity Increase vs Quanser: Sets of Gains

#### Quanser

 Quanser: operates in only one flight condition hence only has one set of gains.

#### F/A-18 UAV

- F/A-18: operates in a range of flight conditions hence has an infinite set of gains.
- Offline: gains are picked a priori at some finite set of points in the range of flight conditions (altitude, speed).
- Online: gains during runtime are computed by interpolation on the pre-defined offline gains.
- We call this gain-scheduling. The F/A-18 controllers are gain-scheduled at 110 different points.



#### Application to Industrial Example: F/A-18 UAV # Of Stability Analysis

#### Stability Analysis

- Need to generate a stability proof for each controller mode (33) for each set of gains (110).
- Total number of configurations is 3630.



# Pitch Controller (Longitudinal)



### Application to Collision avoidance TCAS / last resort safety net



# Conclusion

- It is possible to generate safety-critical control code from specifications, allequipped with semantics and proofs.
- With that, code-level analyses are possible, and much easier than analyses from code alone.



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- Army Research Office
- Dutton/Ducoffe professorship at Georgia Tech
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- ONERA DCSD and DTIM
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# Analyzing control command software: the need for non linear invariants

Formal methods for Aerospace Applications - FMCAD'12 Tutorial

Pierre-Loïc Garoche – Onera

Mon October 22nd 2012

#### CONTENT

- Need for non linear invariants
- Proving stability at code level

### • Non linear invariant synthesis

Quadratic invariants for linear systems Linear systems with guards Beyond linear systems: polynomial controllers

#### Conclusion

#### CONTENT

#### • Need for non linear invariants

Proving stability at code level

#### • Non linear invariant synthesis

Quadratic invariants for linear systems Linear systems with guards Beyond linear systems: polynomial controllers

Conclusion

#### NON LINEAR INVARIANTS IN CONTROL COMMAND SOFTWARE

Properties of controllers:

- open-loop stability
- close-loop stability
- tracking
- • •

(Most | All) of them can be expressed as invariants over the system's variables.

 $\implies$  Lyapunov functions

A controller is open-stable

 $\stackrel{\mathbf{i}}{\triangleq} \text{its output stay bounded for any bounded input.}$ Example:  $in_0 \in [-1, 1], in_1 \in [-1, 1]$ 

$$\begin{pmatrix} x_{0} \\ x_{1} \\ x_{2} \\ x_{3} \end{pmatrix} = \begin{pmatrix} 0.6227 & 0.3871 & 0.0102 & 0.3064 \\ -0.3407 & 0.9103 & -0.3388 & 0.0649 \\ 0.0918 & -0.0265 & -0.7319 & 0.2669 \\ 0.2643 & -0.1298 & -0.9903 & 0.3331 \end{pmatrix} \begin{pmatrix} x_{0} \\ x_{1} \\ x_{2} \\ x_{3} \end{pmatrix} + \begin{pmatrix} 0.3064 & 0.1826 \\ -0.0054 & 0.6731 \\ +0.0494 & 1.6138 \\ -0.0531 & 0.4012 \end{pmatrix} \begin{pmatrix} in_{0} \\ in_{1} \end{pmatrix}$$

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#### Theorem

Any stable linear controller admits a quadratic Lyapunov function

Analyzing controllers

- 1. Abstract interpretation
- 2. SMT-based model checking (k-induction)
- 3. Deductive methods (Weakest precondition)

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**Analyzing controllers** 

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SMT based model-checking: encodes the system states in SMT and the operational semantics as SMT predicates: I(x) and T(x,y)

Deductive methods manipulate logical expressions (eg. SMT) and transform them according to the program semantics.

Both techniques reason on formulas based on the program/model axiomatisation in SMT.

Nowadays very few non linear reasoning at SMT level

#### CONTENT

Need for non linear invariants

#### Proving stability at code level

• Non linear invariant synthesis Quadratic invariants for linear systems Linear systems with guards Beyond linear systems: polynomial controllers

Conclusion

#### **PROVING STABILITY AT CODE LEVEL**

- input: C code annotated with ACSL describing quadratic invariants
- goal: prove the validity of each Hoare triple



#### AN ELLIPSOID-AWARE HOARE LOGIC

• To use ellipsoids to formally specify bounded input, bounded state stability in.
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- To use ellipsoids to formally specify bounded input, bounded state stability in.
- Typically, an instruction *S* would be annotated in the following way:

$$\{x \in \mathcal{E}_P\} \ y = Ax + b \ \{y - b \in \mathcal{E}_Q\}$$
(1)

where the pre- and post- conditions are predicates expressing that the variables belong to some ellipsoid, with  $\mathcal{E}_p = \{x : \mathbb{R}^n | x^T P^{-1} x \leq 1\}$  and  $Q = APA^T$ .

## AN ELLIPSOID-AWARE HOARE LOGIC

The mathematical theorem that guarantees the relations is :

Theorem If *M*, *Q* are invertible matrices, and  $(x - c)^T Q^{-1}(x - c) \leq 1$  and y = Mx + bthen  $(y - b - Mc)^T (MQM^T)^{-1}(y - b - Mc) \leq 1$ 

We will refer to it as the *ellipsoid affine combination theorem*.

- Predicate transformer + external automatic decision procedure (e.g. SMT solver)
- Weakest precondition computation:  $Pre \implies WP(Code, Post)$
- expressiveness of predicate vs. power of decision procedures

In our case:

- code: linear controller
- Post, Pre: quadratic expressions

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In-the-middle annotations act as proof cuts.

Extension of ACSL to manipulate

- matrices, vectors
- properties over matrices
- ellipsoids
- link between C variables and matrices/vectors

ACSL

//@ type matrix; type vector

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/ /	/@ type	e matrix; type vector	ACSL
a	logic	real mat select(matrix A, integer i, integer	ACSL
<u></u>	logic	<pre>integer mat_row(matrix A);</pre>	- <i>ו</i> , <i>ו</i>
Q	logic	<pre>integer mat_col(matrix A);</pre>	

Extension of ACSL to manipulate

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inverse of a matrix *A*, mat\_inverse(*A*) is defined using the predicate is\_invertible(*A*) as follows:

```
/*@ axiom mat_inv_select_i_eq_j:
@ \forwatrixA, integer i, j;
@ is_invertible(A) && i == j ==>
@ mat_select(mat_mult(A, mat_inverse(A)), i, j) = 1
@
@ axiom mat_inv_select_i_dff_j:
@ dxiom mat_inv_select_i_dff_j:
@ forwatrixA, integer i, j;
@ is_invertible(A) && i! = j ==>
@ mat_select(mat_mult(A, mat_inverse(A)), i, j) = 0
@*/
```

Extension of ACSL to manipulate

- matrices, vectors
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Complex constructions or relations can be defined as uninterpreted predicates.

• The following predicate is meant to express that vector x belongs to  $\mathcal{E}_{\mathcal{P}}$ :

//@ predicate in\_ellipsoid(matrix P, vector x);

ACSL

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ACSL

 mat\_of\_array or vect\_of\_array, is used to associate an ACSL matrix type to a C array.

//@ logic matrix mat\_of\_array{L}(float \*A, integer ro integer col);

# A PVS LIBRARY FOR ELLIPSOIDS

A NASA PVS Library to manipulate

- matrices, vectors
- ellipsoids
- affine combination of ellipsoids (thm1)
- S-procedure (thm2)

PVS
Mapping:TYPE= [# dom: posnat, codom: posnat, mp:
[Vector[dom]->Vector[codom]] #]
PVS
$L(n,m)(f) = (\# rows:=m, cols:=n, matrix:=\lambda(j,i):$
f`mp(e(n)(i))(j) #)
$T(n,m)(A) = (\# \text{ dom}:=n, \text{ codom}:=m, mp:=\lambda(x,j): \sum_{i=0}^{A^{\circ}\text{cols}-1}(\lambda(i):$
A`matrix(j,i)*x(i) #))
PVS
Matrix_inv(n):TYPE = {A: Square   squareMat?(n)(A) and
bijective?(n)(T(n,n)(A))}
PUS
inv(n)(A) = L(n,n)(inverse(n)(T(n,n)(A)))

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Vector\_no\_param: TYPE = [# length: posnat, vect: vectors[length].Vector #] PVS

PVS

```
in_ellipsoid?(P: Matrix, x:Vector_no_param ):
MACRO bool =
IF x'length = P'cols AND P'cols=P'rows
THEN ((x'vect)*(P*(x'vect)) <=1)
ELSE FALSE
ENDIF
```

#### **A PVS** LIBRARY FOR ELLIPSOIDS

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```
ellipsoid_affine_comb: LEMMA \forall (n:posnat, Q, M:
SquareMat(n), x, y, b, c: Vector[n]):
bijective?(n)(T(n,n)(Q)) AND bijective?(n)(T(n,n)(M))
AND (x-c)*(inv(n)(Q)*(x-c)) \leq 1
AND y=M*x + b
IMPLIES
(y-b-M*c)*(inv(n)(M*(Q*transpose(M)))*(y-b-M*c)) \leq 1
```

# USING FRAMA-C/JESSIE/WHY DO GENERATE PVS PROOF OBJECTIVE



# Generating PVS PO with Frama-C/Jessie/Why:

- www.frama-c.com (open source C code analysis framework)
- axiomatize C semantics into Why (Jessie)
- WP computation (Why)
- PVS backend to express PO



#### **THEORY INTERPRETATION: MAPPING PVS CONCEPTS**

Theory interpretation is a logical technique for relating one axiomatic theory to another.

```
IMPORTING acsl_theory{{ matrix := Matrix,
vector := Vector_no_param,
vect_length := LAMBDA (v:Vector_no_param): v'length,
mat_row := LAMBDA (M:Matrix): M'rows,
mat_col := LAMBDA (M:Matrix): M'cols,
mat_mult := *,
in_ellipsoid := in_ellipsoid?
mat_inv := LAMBDA (M:Matrix): IF square?(M) THEN IF
bijective?(M'rows)(T(M'rows,M'rows)(M))
THEN inv(M'rows)(M)
ELSE M
ENDIF
ELSE M ENDIF }}
```

# **REFORMULATED PO**



For both POs,

• we must first interpret the uninterpreted types and to prove the properties that are defined axiomatically.

# **REFORMULATED PO**



For both POs,

- we must first interpret the uninterpreted types and to prove the properties that are defined axiomatically.
- We must then discharge the verification conditions. This is done by using PVS and our linear algebra extension of it.

• We have described a global approach to validate stability properties of C code implementing controllers.

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- Need for non linear invariants
- Proving stability at code level

#### Non linear invariant synthesis

Quadratic invariants for linear systems Linear systems with guards Beyond linear systems: polynomial controllers

Conclusion

# **Providing tools to manipulate/synthesize non linear invariants**

Mathematical tools exist to deal with non linear arithmetic:

- Linear systems admit quadratic invariants: use semi definite programming with LMIs
- Polynomial systems admitting polynomial invariants: use Bernstein polynomials
- More complex systems: other tools

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Mathematical tools exist to deal with non linear arithmetic:

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- More complex systems: other tools

Be careful: most of them rely on floating point computations

 $\implies$  find ways to validate the result

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# **QUADRATIC INVARIANTS**

Goal: characterize an ellipsoid such that it is a Lyapunov function for the system

Control theorists rely on LMI and semi definite programming to generate Lyapunov functions.

Let's do the same.

#### **Overall Method**

- 1. First determine the *shape* of the ellipsoid by choosing a matrix *P* such that  $A^T P A P \prec 0$ .
- 2. then find the smallest possible *ratio*  $\lambda$  such that  $x^T P x \leq \lambda$  is an invariant.





# **METHODOLOGY - MULTIPLE APPROACHES**

**Definition (Semidefinite Programming)** 

Minimize a linear objective function of variables  $y_i$  under constraint

$$A_0 + \sum_{i=1}^k y_i A_i \succeq 0$$

where the  $A_i$  are known matrices and " $P \succeq 0$ " means  $x^T P x \ge 0$  for all vector x.

Heuristics

1. Minimizing Condition Number: finding the roundest possible solution

 $I \leq P \leq rI.$ 

2. Preserving the shape: minimizing *r* s.t.

 $A^T P A - rP \preceq 0.$ 

3. Handle the inputs (avoid the scale phase)

#### **ROUNDING ERRORS**

Actual computations are carried out with floating point numbers leading to *rounding errors*.

#### Example

int i = 0; float x = 0; while (i < 1000000) { x += 0.1; ++i; } printf("%.0f\n", x);

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We have to distinguish two problems:

- rounding errors *in the analyzed program*;
- and rounding errors *in the analyzer* itself.

#### **ROUNDING ERRORS IN THE PROGRAM**

Knowing the precision of the floating point system used and the dimensions of matrices *A* and *B* of the analyzed system, we can compute two reals *a* and *b* such that if

$$(Ax + Bu)^T P (Ax + Bu) \leqslant \lambda$$

then

$$\operatorname{fl}(Ax + Bu)^T P \operatorname{fl}(Ax + Bu) \leq a^2 \lambda + 2ab\sqrt{\lambda} + b^2$$

with fl(*e*) the computation of *e* in any order and with any IEEE754 rounding mode (in practice *a* is near from 1 and *b* from 0).
#### **SOUNDNESS OF THE RESULT**

- Checking the soundness of the result basically amounts to *checking positive definiteness* of a matrix.
- This is done by carefully bounding the rounding errors in a *Cholesky decomposition*.
- Hence an efficient soundness check (in  $O(n^3)$  for an  $n \times n$  matrix).

# **EXPERIMENTAL RESULTS**

	Shape	Bounds		Valid.
Ex. 1 n=2, 1 input	0.07	[140.4; 189.9]	0.40	0.01
	0.16	[22.2; 26.5]	0.28	0.01
	0.23	[16.2; 17.6]	0.20	0.01
Ex. 2 n=4, 1 input	0.09	[18.1; 25.2; 24.3; 33.7]	0.40	0.01
	0.27	[6.3; 7.7; 2.2; 3.4]	0.27	0.02
	0.40	[1.7; 2.0; 2.2; 2.5]	0.21	0.01
Ex. 3 lead-lag	0.07			$\perp$
controller	0.17	[36.2; 36.1]	0.33	0.01
n=2, 1 input	0.20	[38.8; 20.3]	0.20	0.01
Ex. 4 LQG	0.09	[1.2; 0.9; 0.5]	0.32	0.02
regulator	0.19	[0.9; 0.9; 0.9]	0.26	0.01
n=3, 1 input	0.24	[0.7; 0.4; 0.3]	0.22	0.02

Analysis times (in s) and bounds compared for the three heuristics.

# EXPERIMENTAL RESULTS, CONTINUED

	Shape	Bounds		Valid.
Ex. 5 coupled	0.09	[9.8; 8.9; 11.0; 16.8]	0.43	0.03
mass system	0.24	[5.7; 5.6; 6.4; 10.1]	0.33	0.03
n=4, 2 inputs	0.48	[5.0; 4.9; 4.8; 4.7]	0.22	0.03
Ex. 6 Butterworth	0.10	[7.5; 8.7; 6.1; 7.0; 6.5]	0.38	0.03
low-pass filter	0.32	[3.6; 5.0; 4.7; 8.1; 8.9]	0.29	0.02
n=5, 1 input	0.78	[2.3; 1.1; 1.9; 2.0; 2.9]	0.24	0.03
Ex. 7 Dampened	0.07	[1.7; 2.1]	0.23	0.01
oscillator	0.15	[2.0; 2.0]	0.20	$\perp$
n=2, no input	0.27	[1.5; 1.5]	0.16	0.01
Ex. 8 Harmonic	0.08	[1.5; 1.5]	0.23	0.01
oscillator	0.24	[1.5; 1.5]	0.20	$\perp$
n=2, no input	0.15	[1.5; 1.5]	0.16	0.01

Analysis times (in s) and bounds compared for the three heuristics.

#### EXPERIMENTAL RESULTS, CONTINUED



Figure: Comparison of obtained ellipsoids by the three methods from lighter to darker, plus a random simulation trace ((b) and (d), being of dimension greater than 2, are cuts along planes containing the origin and two vectors of the canonical base, to show how the three different templates compare together).

#### EXPERIMENTAL RESULTS, END



Figure: Comparison of obtained ellipsoids by the three methods from lighter to darker, plus a random simulation trace (a) and (b), being of dimension greater than 2, are cuts along planes containing the origin and two vectors of the canonical base, to show how the three different templates compare together).

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# LINEAR SYSTEMS WITH GUARDS: EXTENSION TO POLICY ITERATION

Real systems are not purely linear, they use saturations.

**Extension to policy iterations** 

- Build the control flow graph of the program
- Rely on previous approach to compute a set of appropriate templates
- Iterate on program policies with synthesized templates.

$$0,9 \le in \le 1 , x := 10in - 9 \\ y \le 1 , id \\ 0 \le y \le 1 , id \\ 0 \le in < 0,9 , x := 0.2x - 0.7y + 0.5in \\ y := 0.7x + 0.2y + 0.5in$$

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## **BEYOND LINEAR SYSTEMS**

Controllers are rarely linear:

- use of trigonometric functions, exponential ...
- use of polynomials

In static analysis, most complex domains only deal with quadratic properties, using semi definite programming.

Proposal: use Bernstein polynomials to bound polynomial templates.

#### **POLYNOMIAL TEMPLATE DOMAINS**

#### Definition

Given a set  $P = \{p_1, \ldots, p_k\}$  of  $k \in \mathbb{N}$  polynomials over  $n \in \mathbb{N}$  variables, an abstract value is defined as a tuple  $(b_1, \ldots, b_k) \in \mathbb{R}^k$ .

Concretization

$$\gamma (b_1, \ldots, b_k) = \begin{cases} (x_1, \ldots, x_n) \in \mathbb{R}^n & p_1 (x_1, \ldots, x_n) \leq b_1 \\ \wedge \ldots \wedge & p_k (x_1, \ldots, x_n) \leq b_k \end{cases} \end{cases}.$$



## **B**ERNSTEIN POLYNOMIALS AS A POLYNOMIAL TEMPLATE DOMAIN ENGINE

Approach: express program semantics as an optimization problem

 $\max \left\{ p\left(x_{1},\ldots,x_{n}\right) \mid q_{1}\left(x_{1},\ldots,x_{n}\right) \leqslant b_{1} \land \ldots \land q_{k}\left(x_{1},\ldots,x_{n}\right) \leqslant b_{k} \right\}$ 

#### **Bernstein polynomials**

• Bernstein basis:

$$B_{i,n}(x) = \binom{n}{i} x^i (1-x)^{n-i}$$

• Polynomials: every polynomial can be expressed in Bernstein basis

$$p=\sum_{i=0}^n b_{p,i}B_{n,i}.$$

• Bound properties: For any polynomial p, for all  $x \in [0, 1]$ ,

$$\min \left\{ b_{p,i} \mid 0 \leqslant i \leqslant n \right\} \leqslant p(x) \leqslant \max \left\{ b_{p,i} \mid 0 \leqslant i \leqslant n \right\}.$$

#### A TEMPLATE ABSTRACT DOMAIN

Express each program construct as an optimization problem.

• Guards

$$[[r(x_1, ..., x_n) \leq 0]]^{\sharp} (b_1, ..., b_k) = (b'_1, ..., b'_k)$$
  
with, for  $i \in [[1, k]]: b'_i = \max \left\{ p_i(x_1, ..., x_n) \middle| \begin{array}{c} p_1(x_1, ..., x_n) \leq b_1 \\ \wedge ... \wedge \\ p_k(x_1, ..., x_n) \leq b_k \wedge \\ r(x_1, ..., x_n) \leq 0 \end{array} \right\}$ 

Assignments

$$\llbracket x_{i_0} := r(x_1, \ldots, x_n) \rrbracket^{\sharp} (b_1, \ldots, b_k) = (b'_1, \ldots, b'_k)$$

with, for 
$$i \in [\![1,k]\!]$$
:  
 $b'_i = \max \left\{ p_i[x_{i_0} \leftarrow r(x_1, \dots, x_n)](x_1, \dots, x_n) \middle| \begin{array}{c} p_1(x_1, \dots, x_n) \leqslant b_1 \\ \land \dots \land \\ p_k(x_1, \dots, x_n) \leqslant b_k \end{array} \right\}$ 

#### **EXAMPLE**

```
\begin{array}{l} x := 0; \, y := \textit{?(0, 0.5)};\\ \text{while } x \leqslant 1 \ \text{do} \\ y := y + 0.001 \times (18x^2 - 18x + 3);\\ x := x + 0.001;\\ \text{if } y \leqslant 0 \ \text{then } y := 0 \ \text{else } y := y \ \text{fi} \\ \text{od} \end{array}
```



## **Soundness of the result**

All computation were done in floats

PVS NASA library for Bernstein polynomials:

Checking that the floating point result is a sound maximum value in real computations.

See NASA Langley Grizzly

## CONTENT

- Need for non linear invariants
- Proving stability at code level

• Non linear invariant synthesis Quadratic invariants for linear systems Linear systems with guards Beyond linear systems: polynomial controllers

#### Conclusion

#### **SUMMARY**

Identified need for avionics software: non linear reasoning Proposals:

- Proof-assistant that is able to discharge proofs about ellipsoids
  - \* Backend of the Geneauto translation from Simulink + Proof to C code
  - \* Targeting a fully automatic proof replay at C code level
- Abstract domains building non linear abstract values
  - \* quadratic invariants for linear controllers (automatic)
  - \* quadratic invariants for linear controllers with guards (automatic)
  - \* polynomial invariants for polynomial controllers (need to be provided with templates)

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